

FAA APPROVED
AIRCRAFT FLIGHT MANUAL SUPPLEMENT
For
SYSTEM INSTALLATION
Of
EIS-61000 ELECTRONIC IGNITION SYSTEMS

Registration No. _____

Serial No. _____

This supplement must be attached to the FAA approved Aircraft Flight Manual when the Electroair electronic ignition system has been installed per FAA STC #SA03286CH.

The information contained herein supplements or supersedes the basic Aircraft Flight Manual only in those areas listed. For limitations, procedures, and performance information not contained in this supplement, consult the basic Aircraft Flight Manual.

FAA APPROVED: _____
for Monica Merritt Manager, Southwest Flight Test Section, AIR-713
Federal Aviation Administration
Ft. Worth, TX

This document has been designated as Electroair Restricted Proprietary Information.
The information contained therein is reserved for the exclusive use of Electroair.
Release only on a need to know basis.

ELECTROAIR RESTRICTED

Log of Revisions:

Revision	Pages Affected	Date of Revision	Description of Revision	Approved by	Date of FAA Approval
00		01/08/2014	Initial Release	RDM	01/13/2014
01		06/09/2014	Updated document number.	RDM	08/06/2014
02		08/12/2014	ECO 1116-0069		
03		10/09/2014	ECO 1116-0082	RDM	01/09/2015
04		01/16/2015	ECO 1116-0108		
05		08/10/2015	ECO 1116-0135		
06	4, 12	03/20/2018	ECO 1019-0051	<u>EJN</u>	<u>6/18/2018</u>
07	4, 12	05/17/2019	SA09950CH-A		
08	2, 4-5, 12-13	05/20/2021	SA09966CH-A		

Table of Contents:

Log of Revisions:	2
Section 1: General:	4
Section 2: Limitations:	6
Section 3: Emergency/Abnormal Procedures:	7
Emergency Procedures.....	7
Abnormal Procedures.....	7
Section 4: Normal Procedures:	8
I. Preflight:	8
II. Starting:.....	8
III. Ignition Check:.....	8
IV. Takeoff:.....	10
V. Cruise:	10
VI. Descent:.....	10
VII. Landing:	10
VIII. Shutdown:.....	10
IX. Post-flight:.....	10
Section 5: Performance:	11
Section 6: Weight and Balance:	12
Single CSTW Kit Continental	12
Single CSTW Kit Lycoming	12
Single MTH Kit Continental/Lycoming	12
Section 7: Glossary and Abbreviations:	13

Section 1: General:

This Aircraft Flight Manual Supplement contains the necessary information required for the operation of an electronic ignition system (EIS) as installed on 6 cylinder Continental E-Series, 300, 360, 470, 520, and 550 naturally aspirated and turbocharged series engines along with Lycoming 540, 541, and 580 naturally aspirated and turbocharged series engines.

A. FUNCTIONAL OVERVIEW

The Electroair EIS-61000 Electronic Ignition System is a single magneto replacement. An aircraft equipped with an EIS-61000 and a single magneto will make up the dual ignition system. The EIS-61000 kit consists of the following components: Controller (EA-21000), Coil Pack (EA-8000), Spark Plug Wires (EA-4000T), Wire Harness (EA-22000), and Trigger Mechanism (EA-11000, EA-9000A, EA-10000, EA-12000, EA-16000, EA-17000 or EA-19000).

The EIS-61000 Electronic Ignition System performs its function by delivering energy generated by the coil pack to each spark plug attached to the system. This high voltage from the coil pack (on the order of 70,000V), creates a high intensity, long duration spark which more effectively ignites a wide range of fuel/air mixtures inside of the cylinder. The EIS-61000 is also able to vary the ignition timing (spark event) during the combustion cycle so as to more closely have the peak pressure as a result of combustion occur at an optimal range for a piston engine. The adjustment of ignition timing is based on MAP inside the engine. The combination of a high energy spark and variable timing, the two principle differences between the EIS-61000 and a magneto, permits more an efficient operation of the engine.

The EIS-61000 is operated by DC power provided by the aircraft's power bus. There are two circuit protection devices used for the EIS-61000; reference table one for the type and size of the protection devices. These circuit protection devices are not normally accessed during flight.

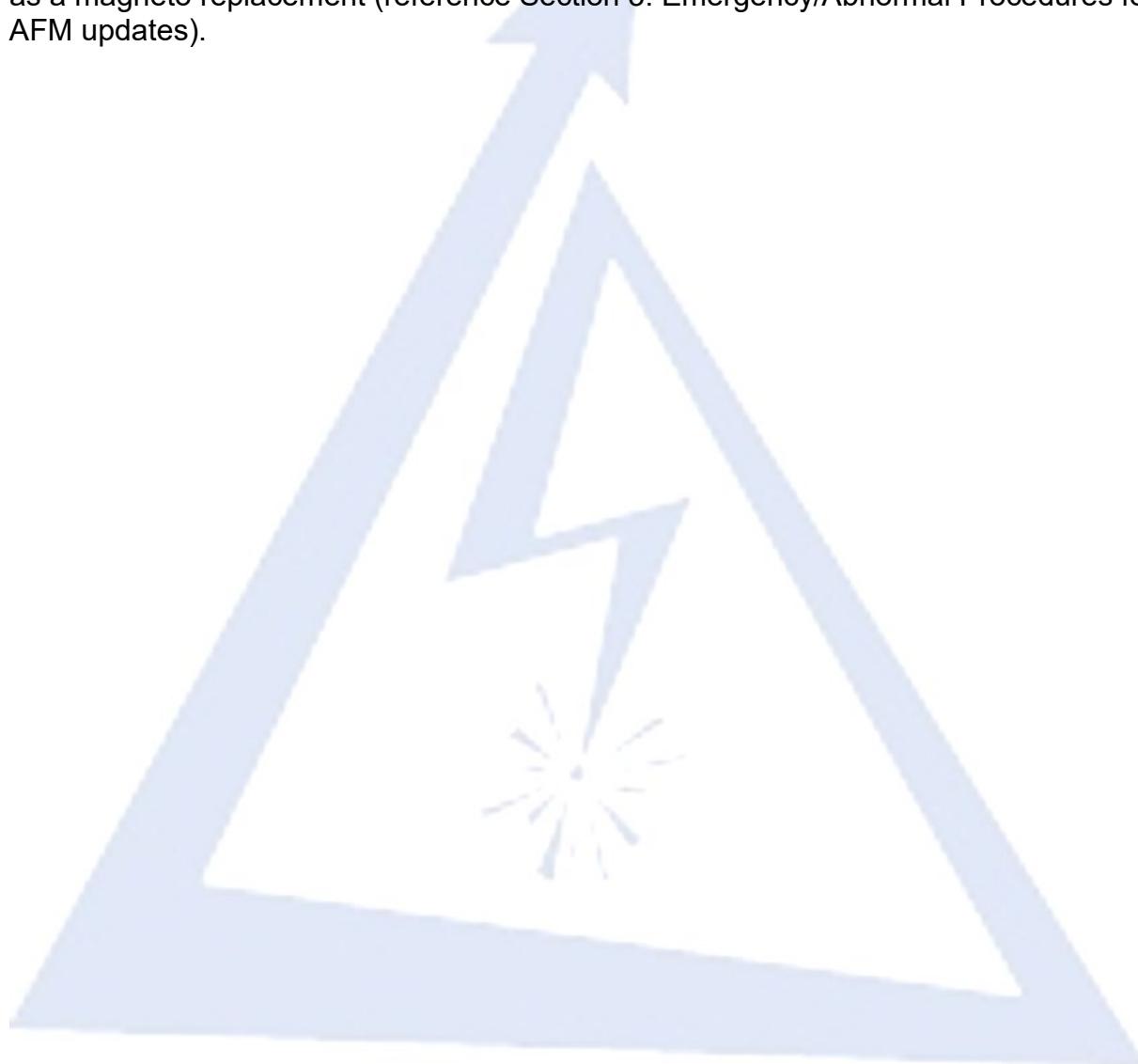
The EIS-61000 is controlled by using the switch labeled "EIS". The EIS-61000 may be disabled by setting the switch labeled "EIS" to the OFF position. Pilots should familiarize themselves with the location of the "EIS" before proceeding with the pre-flight checklist.

FUNCTION	IDENT	RATING	BUS	POWER SUPPLY
EA-21000 Power	EIS	2.0 AMP	Aircraft Power	12/24VDC
EA-8000 Power	EIS	10.0 AMP	Aircraft Power	12/24VDC

Table One: DC Circuit Protection

B. SYSTEM OPERATION

Under normal operating conditions, the EIS-61000 Electronic Ignition System will be controlled by the flight crew in the same manner as the magneto that was previously installed. The AFM shall be updated by adding the task of turning ON the system (EIS) after the Master Switch is turned ON; and by turning OFF the system (EIS) prior to the Master Switch being turned OFF (reference Section 4: Normal Procedures for AFM updates). The AFM shall also be updated in the magneto trouble-shooting section to change wording to reflect the fact than an Electronic Ignition System has been installed as a magneto replacement (reference Section 3: Emergency/Abnormal Procedures for AFM updates).



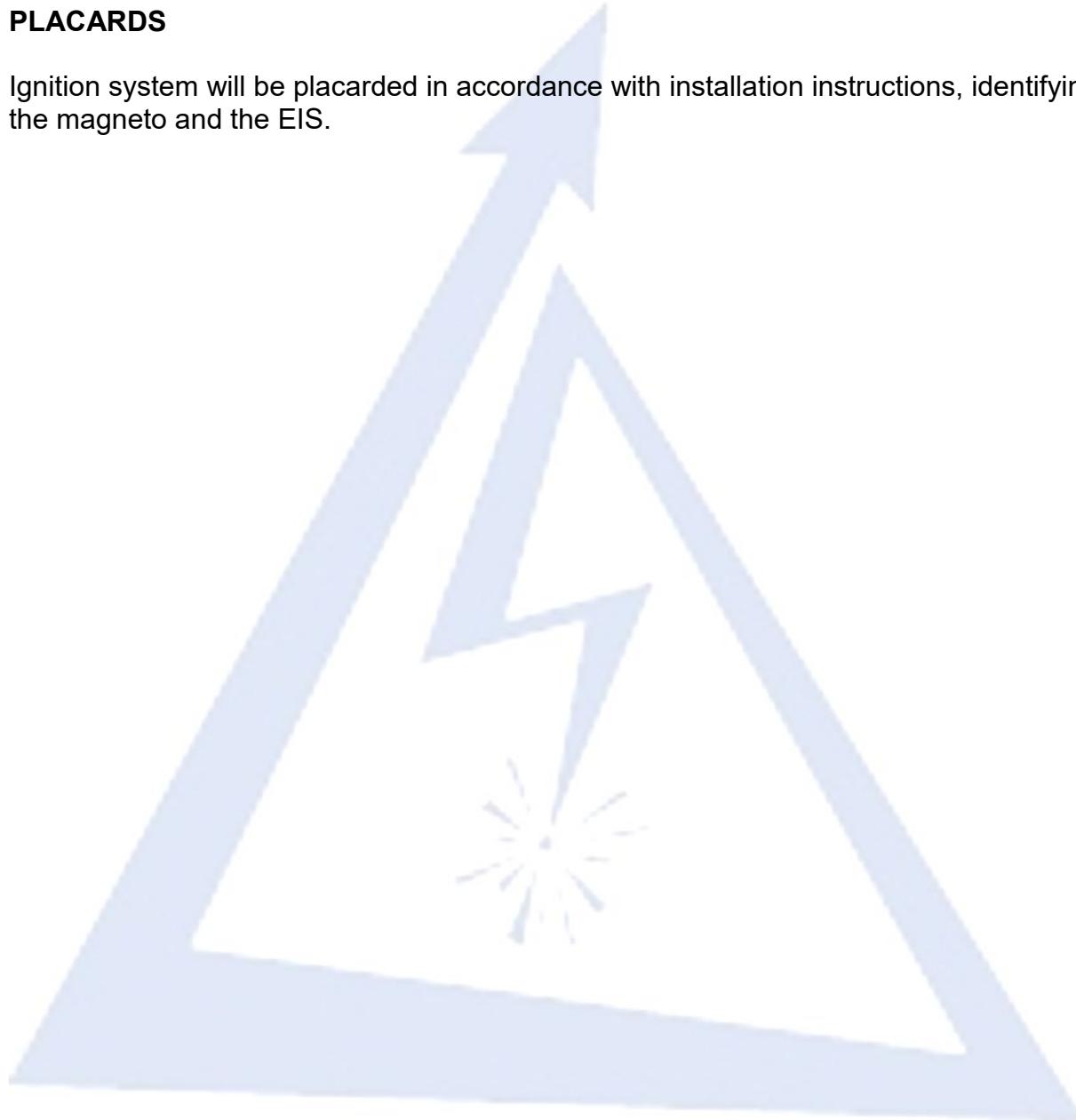
Section 2: Limitations:

VOLTAGE

System voltage to EIS-61000 shall be at least 8VDC, or EIS will not function.

PLACARDS

Ignition system will be placarded in accordance with installation instructions, identifying the magneto and the EIS.



Section 3: Emergency/Abnormal Procedures:

Emergency Procedures

No Change

Abnormal Procedures

Alternator/Generator Failure:

No Change

Note: It is important to take into consideration that the EIS-61000(s) will be drawing power from the aircraft's remaining power supply. For load shedding calculations, use the value of 1.25 Amps for what one EIS-61000 will draw. It is NOT recommended to load shed the EIS.

Problem: Rough running engine and/or high CHTs:

Perform in-flight troubleshooting in accordance with POH. Determine if one of the ignition systems is bad and isolate the bad/suspected ignition system.

Problem: Severe loss in engine power and/or low CHTs, engine operating smoothly:

Perform in-flight troubleshooting in accordance with POH. Determine if one of the ignition systems is bad and isolate the bad/suspected ignition system.

Section 4: Normal Procedures:

I. Preflight:

Update the POH to include turn "EIS-ON" after turn Master Switch ON.

II. Starting:

"EIS Switch – ON" Procedure:

Verify the switch labeled "EIS" to the ON position.

III. Ignition Check:

Procedure, With Rotary or Key Switch:

The ignition check shall be made at the same RPM as defined in the AFM.

Ignition switch to "BOTH" position

Ignition switch to "R" position -- Note RPM Drop _____

Ignition switch to "BOTH"

Ignition switch to "L" position – Note RPM Drop _____

Ignition to "BOTH" position

Magneto RPM drop should not exceed the RPM defined in the AFM. EIS RPM drop is defined below. If there is a doubt concerning operation of the ignition systems, RPM checks at higher engine speeds will usually confirm whether a deficiency exists. At the end of the ignition check move ignition switch back to "BOTH" position.

RPM Drops:

Magneto – as defined in POH.

EIS – 30-40 RPM max

Procedure, With Standard or Rocker Switch:

The ignition check shall be made at the same RPM as defined in the AFM.

Ignition BOTH ON

EIS OFF – Note RPM Drop _____

EIS ON

Magneto OFF – Note RPM Drop _____

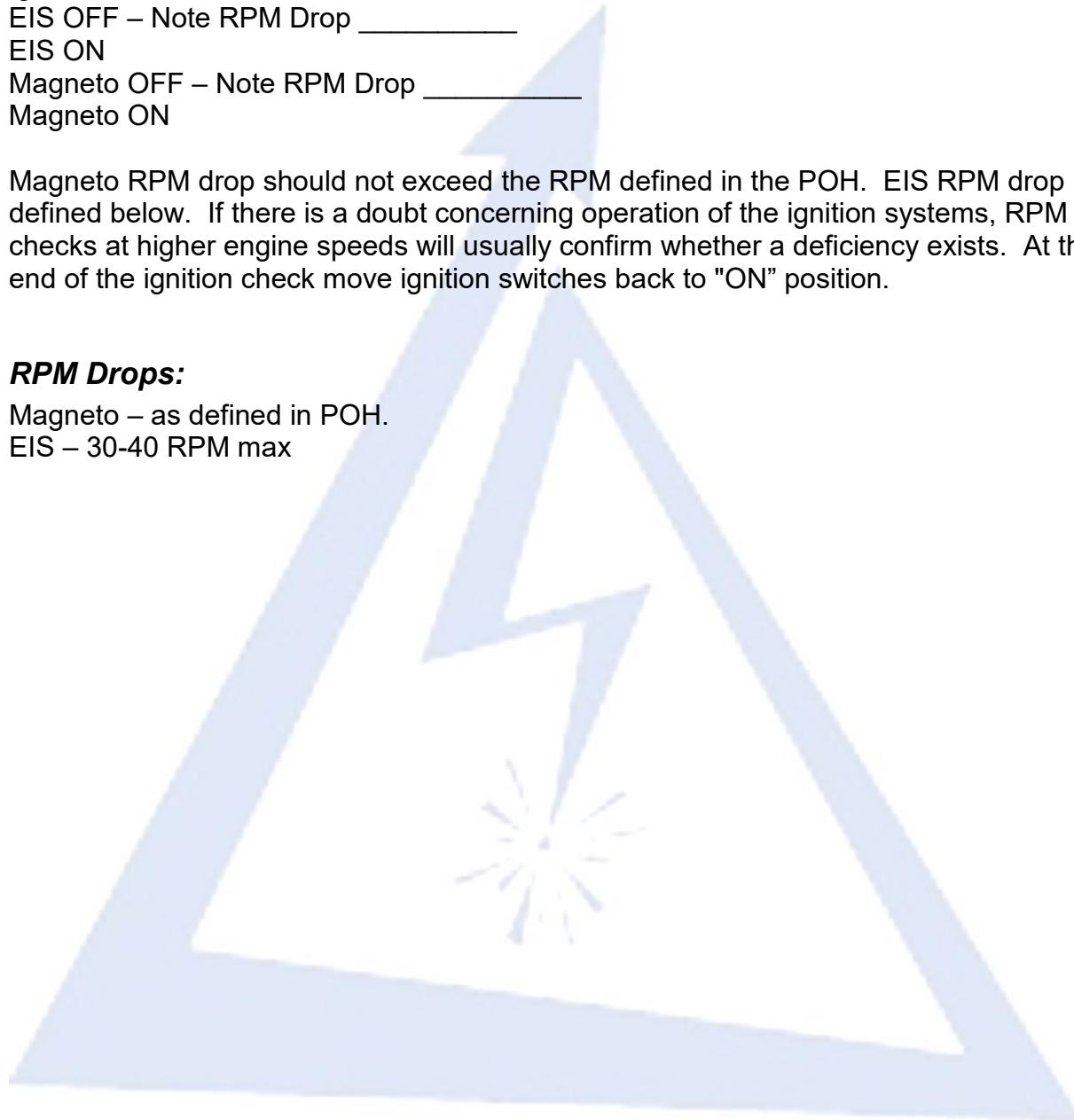
Magneto ON

Magneto RPM drop should not exceed the RPM defined in the POH. EIS RPM drop is defined below. If there is a doubt concerning operation of the ignition systems, RPM checks at higher engine speeds will usually confirm whether a deficiency exists. At the end of the ignition check move ignition switches back to "ON" position.

RPM Drops:

Magneto – as defined in POH.

EIS – 30-40 RPM max



IV. Takeoff:

No Change

V. Cruise:

No Change

VI. Descent:

No Change

VII. Landing:

No Change

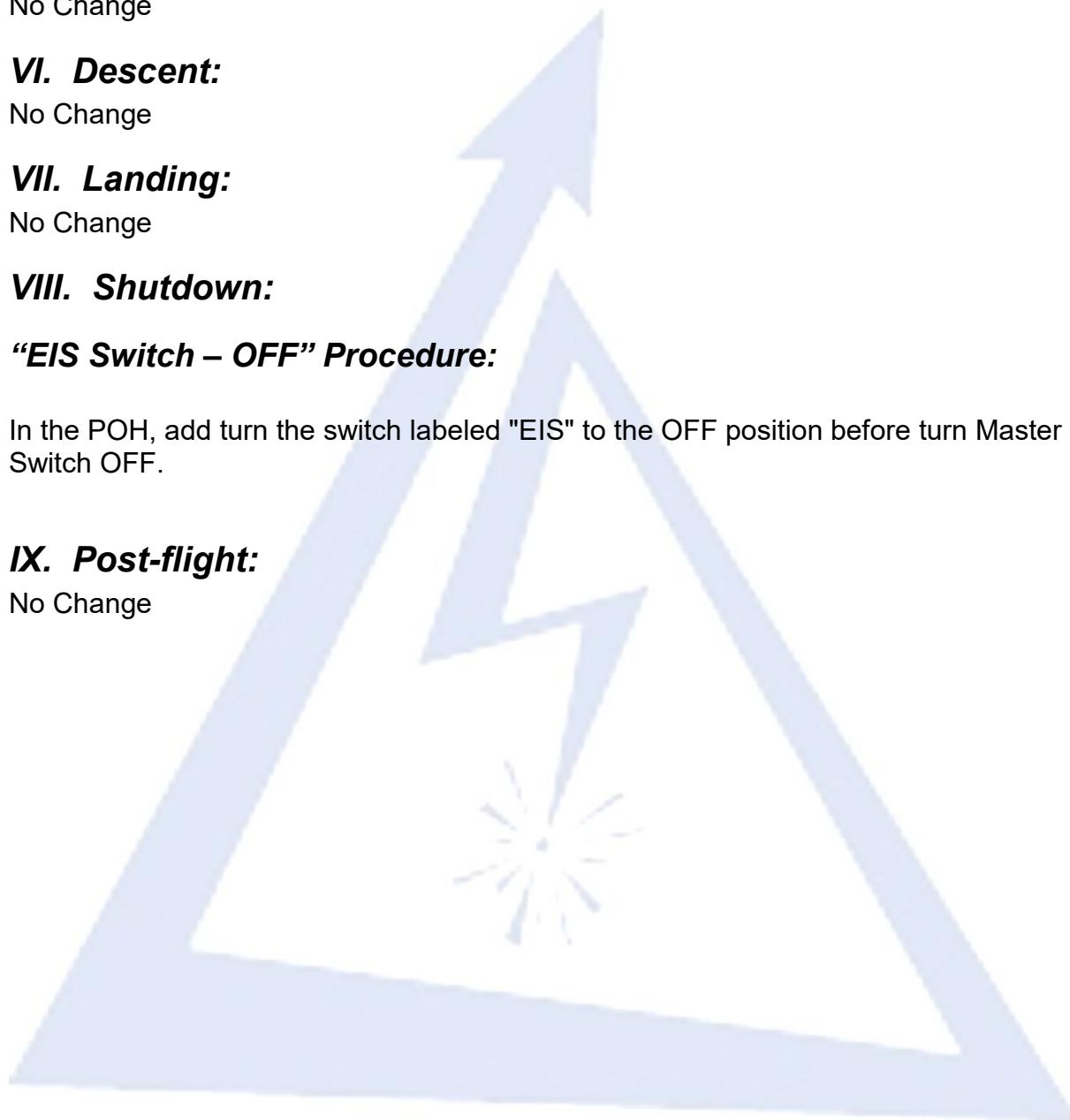
VIII. Shutdown:

“EIS Switch – OFF” Procedure:

In the POH, add turn the switch labeled "EIS" to the OFF position before turn Master Switch OFF.

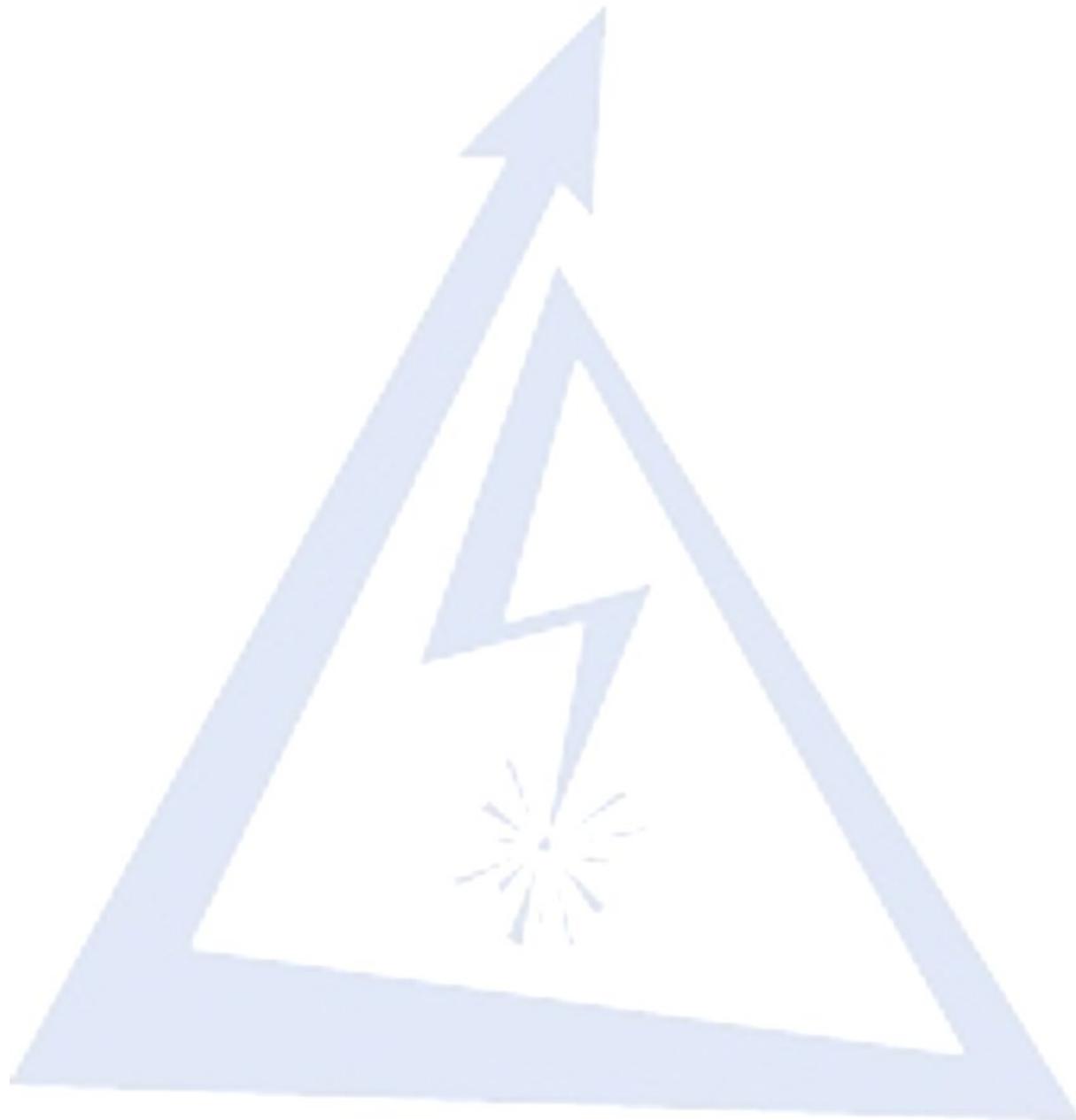
IX. Post-flight:

No Change



Section 5: Performance:

No Changes



Section 6: Weight and Balance:

The installation of the EIS-61000 requires the removal of one Magneto and the installation of the EIS-61000 components. This installation results in a change to the aircraft's weight and balance. A new weight and balance should be calculated for the aircraft after the installation of the EIS-61000(s). All future loading calculations should use the updated aircraft weight and balance. The individual EA part weights are below. NOTE: For the twin-engine aircrafts the weight and balance must include the weights of both sets of EIS-61000 components.

Single CSTW Kit Continental

1. EA-21000: 0.8 pounds (Controller)
2. EA-8000: 3.7 pounds (Coil Pack)
3. EA-11030: 1.0 pound (Magneto Hole Cover & Brackets)
4. EA-4000T: 2.2 pounds (Spark Plug Wires)
5. EA-22000: 0.6 pounds (Controller Wire Harness)
6. EA-11000: 0.4 pounds (CSTW)

Single CSTW Kit Lycoming

1. EA-21000: 0.8 pounds (Controller)
2. EA-8000: 3.7 pounds (Coil Pack)
3. EA-9020: 1.0 pound (Magneto Hole Cover & Brackets)
4. EA-4000T: 2.2 pounds (Spark Plug Wires)
5. EA-22000: 0.6 pounds (Controller Wire Harness)
6. EA-9000A: 0.4 pounds (CSTW)

Single MTH Kit Continental/Lycoming

1. EA-21000: 0.8 pounds (Controller)
2. EA-8000: 3.7 pounds (Coil Pack)
3. EA-4000T: 2.2 pounds (Spark Plug Wires)
4. EA-22000: 0.6 pounds (Controller Wire Harness)
5. EA-10000, EA-12000, EA-16000, EA-17000 or EA-19000: 2.5 pounds (MTH)

Section 7: Glossary and Abbreviations:

AD(s) – Airworthiness Directive(s)
AFM – Aircraft Flight Manual
AFMS – Aircraft Flight Manual Supplement
ALS – Aircraft Limitations Section
AML – Approved Model List
APU – Auxiliary Power Unit
BTDC – Before Top Dead Center
CFR – Code of Federal Regulations
CSTW – Crank Shaft Trigger Wheel
EIS – Electronic Ignition System
FAA – Federal Aviation Administration
Ignition Timing – is the process of setting the angle relative to piston position and crankshaft angular velocity that a spark will occur in the combustion chamber near the end of the compression stroke.
LOPC – Loss Of Power Control
MAG – magneto
MAP – Manifold Absolute Pressure
May/Should – an optional requirement
MEL – Minimum Equipment List
MTH – Mag Timing Housing
Must/Shall – a mandatory requirement
RPM – Revolutions per Minute
POH – Pilot's Operating Handbook
STC – Supplemental Type Certificate
TDC – Top Dead Center